

High order methods in CFD

Discontinuous Galerkin Method

Praveen Chandrashekar

`praveen@math.tifrbng.res.in`

<http://cpraveen.github.io>



Center for Applicable Mathematics
Tata Institute of Fundamental Research
Bangalore-560065, India

<http://math.tifrbng.res.in>

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Finite element Galerkin Method

$$\partial_t \mathbf{U} + \mathbf{R}(\mathbf{U}) = 0$$

- 1 approximate solution in terms of some **basis/trial functions**

$$\mathbf{U}(x, t) \approx \mathbf{U}_h(x, t) = \sum_j \mathbf{U}_j \phi_j(x)$$

- 2 satisfy PDE in a weak manner: make residual orthogonal to all **test functions=trial functions**

$$\int_{\Omega} [\partial_t \mathbf{U}_h + \mathbf{R}(\mathbf{U}_h)] \phi_i(x) dx = 0, \quad \forall \phi_i$$

ϕ_i discontinuous \implies discontinuous Galerkin method

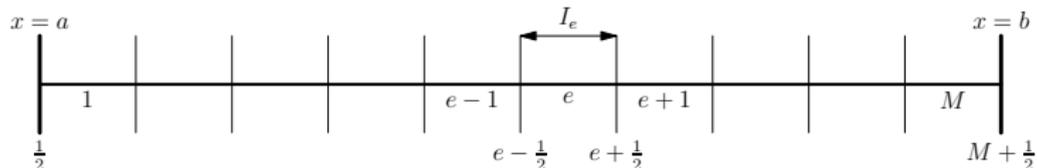
DG for hyperbolic problems

DG in 1-D

Conservation law in 1-D

$$\mathbf{U}_t + \mathbf{F}(\mathbf{U})_x = 0, \quad x \in [a, b], \quad t > 0$$

Partition domain $[a, b]$ into disjoint elements



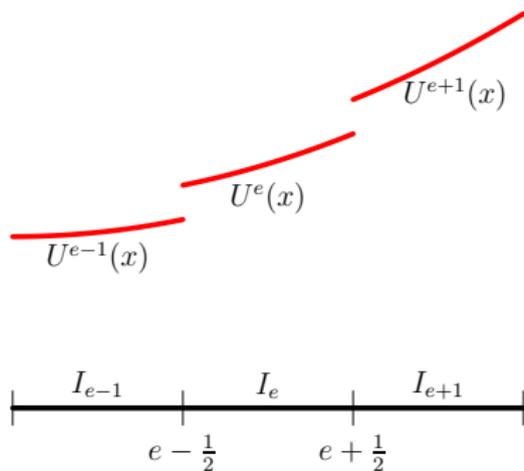
$$I_e = [x_{e-\frac{1}{2}}, x_{e+\frac{1}{2}}], \quad h_e = x_{e+\frac{1}{2}} - x_{e-\frac{1}{2}}, \quad [a, b] = \cup_e I_e$$

Inside each element, approximate solution by polynomial of degree $N \geq 0$

$$x \in I_e : \quad \mathbf{U}_h(x, t) = \mathbf{U}^e(x, t) = \sum_{i=0}^N \mathbf{U}_i^e(t) \phi_i^e(x) \in \mathbb{P}_N(x)$$

$\{\phi_0^e, \phi_1^e, \dots, \phi_N^e\}$ is a basis for \mathbb{P}_N .

The solution U_h is allowed to be discontinuous at the element boundaries.



i.e.,

$$U^e(x_{e+\frac{1}{2}}) \neq U^{e+1}(x_{e+\frac{1}{2}})$$

Galerkin method

$$\int_{I_e} [\mathbf{U}_t + \mathbf{F}(\mathbf{U})_x] \phi_i^e dx = 0$$

Perform integration by parts: $i = 0, 1, \dots, N$

$$\frac{d}{dt} \int_{I_e} \mathbf{U}^e \phi_i^e dx - \int_{I_e} \mathbf{F}(\mathbf{U}^e) \frac{\partial \phi_i^e}{\partial x} dx + (\mathbf{F} \phi_i^e)_{x=x_{e+\frac{1}{2}}} - (\mathbf{F} \phi_i^e)_{x=x_{e-\frac{1}{2}}} = 0$$

Approximate flux at $x = x_{e+\frac{1}{2}}$ by a **numerical flux function** $\hat{\mathbf{F}}$

$$\mathbf{F}(x_{e+\frac{1}{2}}, t) \approx \mathbf{F}_{e+\frac{1}{2}}(t) = \hat{\mathbf{F}}(\mathbf{U}^e(x_{e+\frac{1}{2}}, t), \mathbf{U}^{e+1}(x_{e+\frac{1}{2}}, t))$$

First term is

$$\int_{I_e} \mathbf{U}^e \phi_i^e dx = \int_{I_e} \sum_{j=0}^N \mathbf{U}_j^e \phi_j^e \phi_i^e dx = \sum_{j=0}^N M_{ij}^e \mathbf{U}_j^e$$

Mass matrix $M^e \in \mathbb{R}^{(N+1) \times (N+1)}$

$$M_{ij}^e = \int_{I_e} \phi_i^e \phi_j^e dx$$

System of ODE

$$M^e \frac{dU^e}{dt} + R^e(U^{e-1}, U^e, U^{e+1}) = 0$$

where for $i = 0, 1, \dots, N$

$$R_i^e = - \int_{I_e} \mathbf{F}(U^e) \frac{\partial \phi_i^e}{\partial x} dx + \mathbf{F}_{e+\frac{1}{2}} \phi_i^e(x_{e+\frac{1}{2}}) - \mathbf{F}_{e-\frac{1}{2}} \phi_i^e(x_{e-\frac{1}{2}})$$

Integrate ODE using SSPRK schemes [45]

Popularized in a series of papers by Cockburn & Shu [17], [16], [15], [19]

Some remarks

- 1 Method is conservative
- 2 Numerical fluxes are same as those used in finite volume methods
 - ▶ upwind/Riemann solver based fluxes must be used

- 3 Smooth solutions

$$\|U - U_h\| = O(h^{N+1})$$

- 4 Each element is coupled to its left and right neighbor only, via numerical flux
- 5 Same stencil $\{e - 1, e, e + 1\}$ for all order of accuracy
- 6 Evolve entire polynomial solution, not just cell average
 - ▶ No solution reconstruction required
- 7 High order upto the boundary
 - ▶ no need to change scheme near boundary
- 8 Low dissipation and dispersion errors
- 9 Fixed, compact stencil, high arithmetic intensity: good for HPC

Implementation details: mapped elements, modal basis

Define basis functions on a reference element, e.g., $[-1, 1]$

$$I_e \rightarrow [-1, 1], \quad \xi = \frac{x - x_e}{\frac{1}{2}h_e} \in [-1, 1]$$

Modal basis: basis of orthogonal polynomials

P_i = Legendre polynomial of degree i

$$\phi_i^e(x) = \phi_i(\xi) = \sqrt{2i+1}P_i(\xi)$$

Mass matrix is diagonal

$$M^e = h_e \mathbb{I}_{N+1}, \quad \mathbb{I}_{N+1} = (N+1) \times (N+1) \text{ identity matrix}$$

Example: degree $N = 1$

$$U_h^e = U_0^e + U_1^e \xi = U_0^e + U_1^e \frac{x - x_e}{\frac{1}{2}h_e}$$

$$U_0^e = \text{cell average}, \quad U_1^e = \text{undivided slope}$$

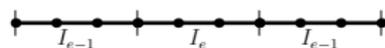
Implementation details: mapped elements, nodal basis

Choose $N + 1$ distinct points in reference element $[-1, 1]$

$$-1 \leq \xi_0 < \xi_1 < \dots < \xi_N \leq +1$$



Basis functions are Lagrange polynomials



$$\phi_i^e(x) = l_i(\xi) = \prod_{j=0, j \neq i}^N \frac{\xi - \xi_j}{\xi_i - \xi_j}, \quad l_i(\xi_j) = \delta_{ij} = \begin{cases} 1 & i = j \\ 0 & \text{otherwise} \end{cases}$$

Coefficients are nodal solution values

$$U^e(\xi) = \sum_{j=0}^N U_j^e l_j(\xi), \quad U_j^e = U^e(\xi_j)$$

Nodes are usually taken to be **Gauss-Legendre** or **Gauss-Lobatto-Legendre** quadrature points

Implementation details: quadrature

- Integrals: need numerical quadrature due to non-linear flux functions
- Optimal accuracy: quadrature rule for flux integral must be exact for polynomials of degree at least $2N$.
- Gauss-Legendre or Gauss-Legendre-Lobatto quadrature use $Q = N + 1$ point rule
- Mass matrix can also be computed by quadrature
- Nodal basis of GL nodes; use same nodes for quadrature
 - ▶ mass matrix is exact (1-D)
 - ▶ mass matrix is diagonal
- Nodal basis of GLL nodes; use same nodes for quadrature
 - ▶ mass matrix is not exact
 - ▶ mass matrix is diagonal

Example: linear advection

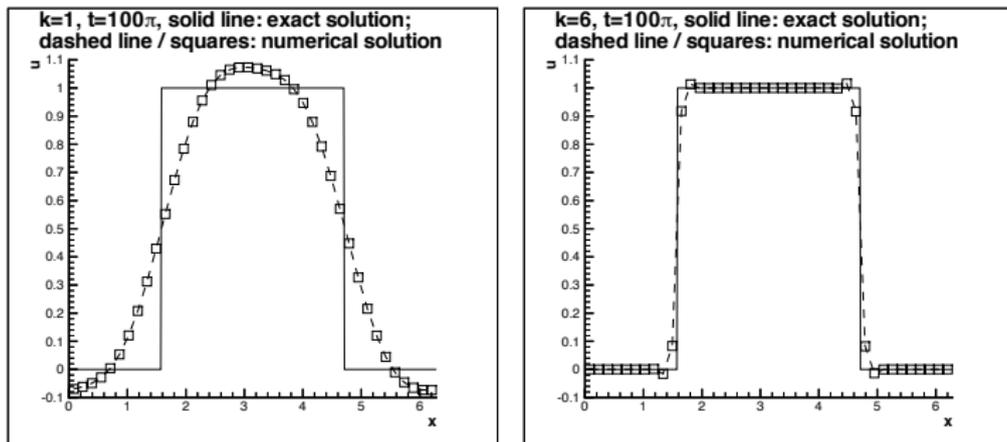


FIG. 2.1. The discontinuous Galerkin method (2.4) applied to the linear equation (2.1) with a square initial condition, $t = 100\pi$. 40 cells. Third order Runge-Kutta in time. Solid line: the exact solution; Dashed line and squares symbols: the computed solution at the cell centers. Left: $k = 1$; Right: $k = 6$.

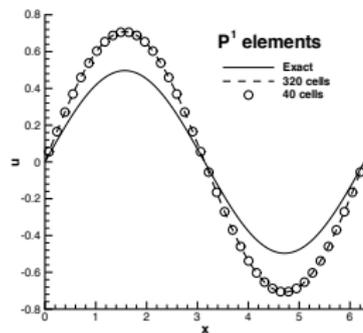
DG for parabolic problems

DG for heat equation: $u_t = u_{xx}$

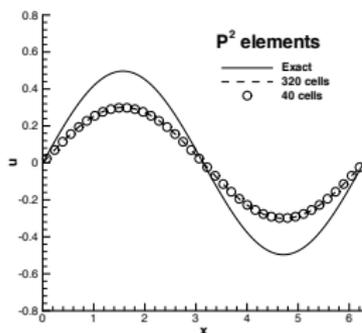
$$\int_{I_e} \phi_i^e \partial_t u_h dx + \int_{I_e} (\partial_x u_h)(\partial_x \phi_i^e) dx + (\phi_i^e \partial_x u_h)_{e-\frac{1}{2}} - (\phi_i^e \partial_x u_h)_{e+\frac{1}{2}} = 0$$

Question: How to approximate heat flux $(\partial_x u_h)_{e+\frac{1}{2}}$?

$$(\partial_x u_h)_{e+\frac{1}{2}} = \frac{1}{2} [(\partial_x u_h)_{e+\frac{1}{2}}^- + (\partial_x u_h)_{e+\frac{1}{2}}^+]$$



(a)



(b)

Converges to wrong solution !!!

Fig. 2.1. The numerically inconsistent discontinuous Galerkin method (2.8) applied to the heat equation (2.1) with an initial condition $u(x,0) = \sin(x)$. $t = 0.7$. Third-order Runge-Kutta in time with small Δt so that time error can be ignored. Numerical solutions with 40 cells (circles) and 320 cells (dashed lines), vs. the exact solution (solid line). (a) $k = 1$; (b) $k = 2$.

DG for parabolic problems

Two methods for stable and accurate schemes

- Interior penalty methods
 - ▶ Primal method, no extra variables
 - ▶ Add extra terms to get stability: symmetric and non-symmetric IP
 - ▶ Elliptic/parabolic: Douglas/Dupont [23], Arnold [2]
 - ▶ Navier-Stokes: Hartmann [34], Chandrashekar [10]¹
- Mixed methods
 - ▶ Write as first order system

$$u_t = q_x, \quad q = u_x$$

Apply DG scheme to this system

- ▶ Both u and q can be computed accurately
- Unified analysis presented in [3]

¹Based on KFVS: see my [slides](#)

DG for parabolic problems: mixed methods

$$u_t = q_x, \quad q = u_x$$

Approximate both u and q using piecewise polynomials

$$u \approx u_h \in \mathbb{P}_N, \quad q \approx q_h \in \mathbb{P}_N$$

Integrate by parts on one element

$$\int_{I_e} \phi_i^e \frac{\partial u_h}{\partial t} dx + \int_{I_e} q_h \frac{\partial \phi_i^e}{\partial x} dx + \phi_i^e(x_{e-\frac{1}{2}}^+) q_{e-\frac{1}{2}} - \phi_i^e(x_{e+\frac{1}{2}}^-) q_{e+\frac{1}{2}} = 0$$

$$\int_{I_e} \phi_i^e q_h dx + \int_{I_e} u_h \frac{\partial \phi_i^e}{\partial x} dx + \phi_i^e(x_{e-\frac{1}{2}}^+) u_{e-\frac{1}{2}} - \phi_i^e(x_{e+\frac{1}{2}}^-) u_{e+\frac{1}{2}} = 0$$

Require numerical fluxes: $u_{e+\frac{1}{2}}, q_{e+\frac{1}{2}}$

BR1 scheme (Bassi & Rebay [5])

$$u_{e+\frac{1}{2}} = \frac{1}{2}(u_{e+\frac{1}{2}}^- + u_{e+\frac{1}{2}}^+), \quad q_{e+\frac{1}{2}} = \frac{1}{2}(q_{e+\frac{1}{2}}^- + q_{e+\frac{1}{2}}^+)$$

- Sub-optimal error: $\|u - u_h\| = O(h^N)$
- Large stencil: neighbour of neighbour
- Odd-even decoupling

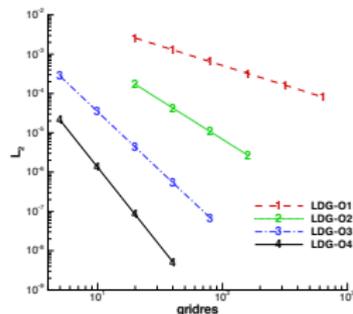
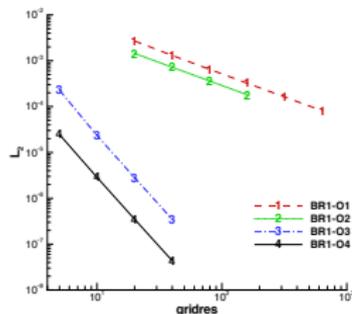
BR2 scheme [9]

- local lifting operator to define face-based flux
- compact stencil: face neighbours only
- optimal order accuracy for all degree

DG for parabolic problems: Numerical fluxes

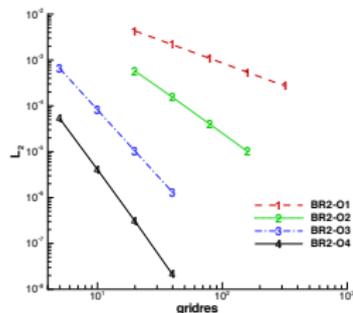
LDG scheme: alternating flux (Cockburn & Shu [18])

$$u_{e+\frac{1}{2}} = u_{e+\frac{1}{2}}^-, \quad q_{e+\frac{1}{2}} = q_{e+\frac{1}{2}}^+$$



- Optimal accuracy:
 $\|u - u_h\| = O(h^{N+1})$ for all N
- Smaller stencil

CDG scheme: (Persson & Peraire [43])



(Landmann [39])

Fourier analysis

Dissipation & dispersion property

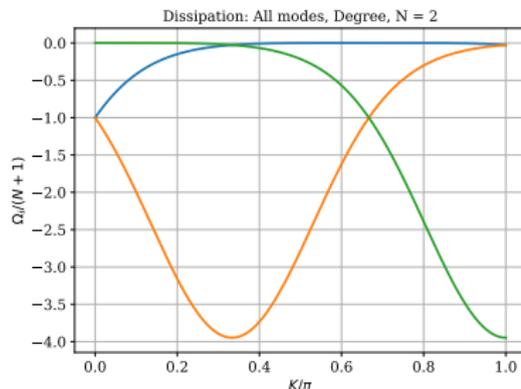
For more details, see my DG [notes](#)

$$U_h^e(x, t) = \hat{U} \exp(i(kx_e - \omega t)) \implies A(kh)\hat{U} = -i\omega h\hat{U}$$

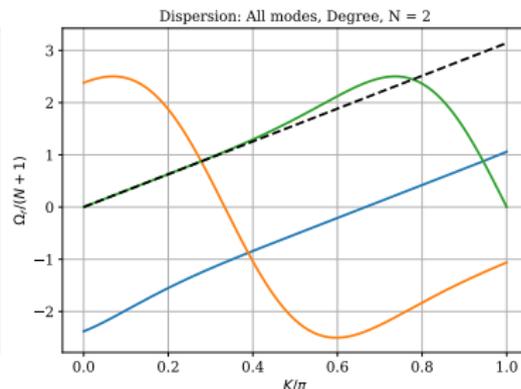
Ideal dispersion relation: $\omega = k$, no dissipation, no dispersion.

Numerical dispersion relation:

Imag(ω) : dissipation

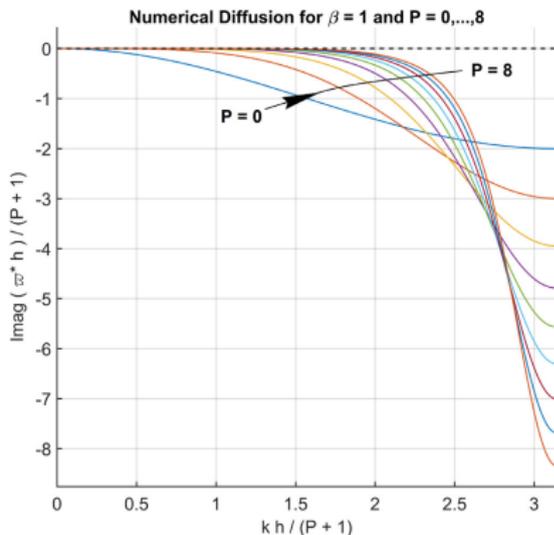
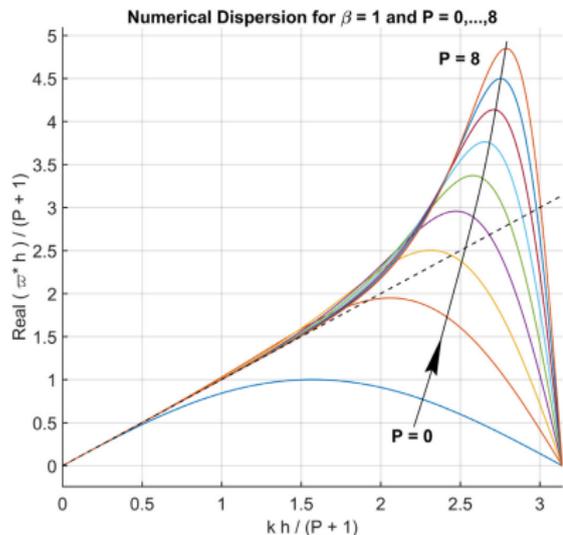


Real(ω) : dispersion



Degree $N = 2$, physical mode is in green

Dissipation & dispersion property



Mengaldo et al. [41]

Dissipation & dispersion property

$$\text{Dispersion: } |Re(\tilde{\omega}^*) - \tilde{k}| < 0.01$$

Maximum resolved wave number and minimum number of unknowns per wavelength for DG schemes, central FD schemes and the DRP scheme.

Scheme	Wave number	Unknowns per wavelength
DG- P^1	0.7716	8.1430
DG- P^2	0.9942	6.3193
DG- P^3	1.1567	5.4318
DG- P^4	1.2754	4.9266
DG- P^5	1.3653	4.6020
FD 2nd order	0.3925	16.0075
FD 4th order	0.7980	7.8733
FD 6th order	1.0841	5.7955
optimized 7-point DRP	1.2469	5.0390

(Cheng & Shu [14])

Dissipation & dispersion property

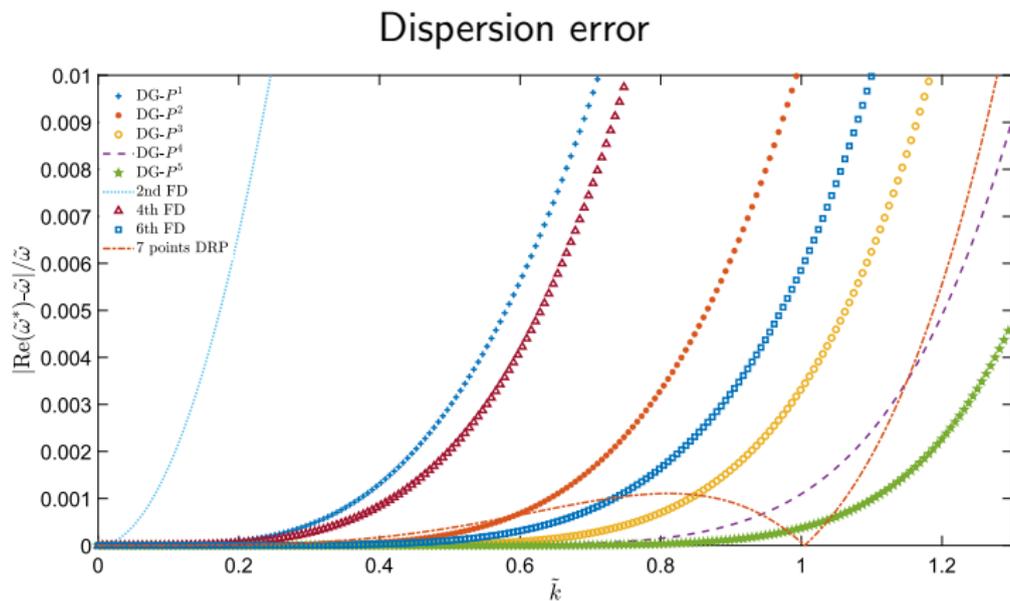
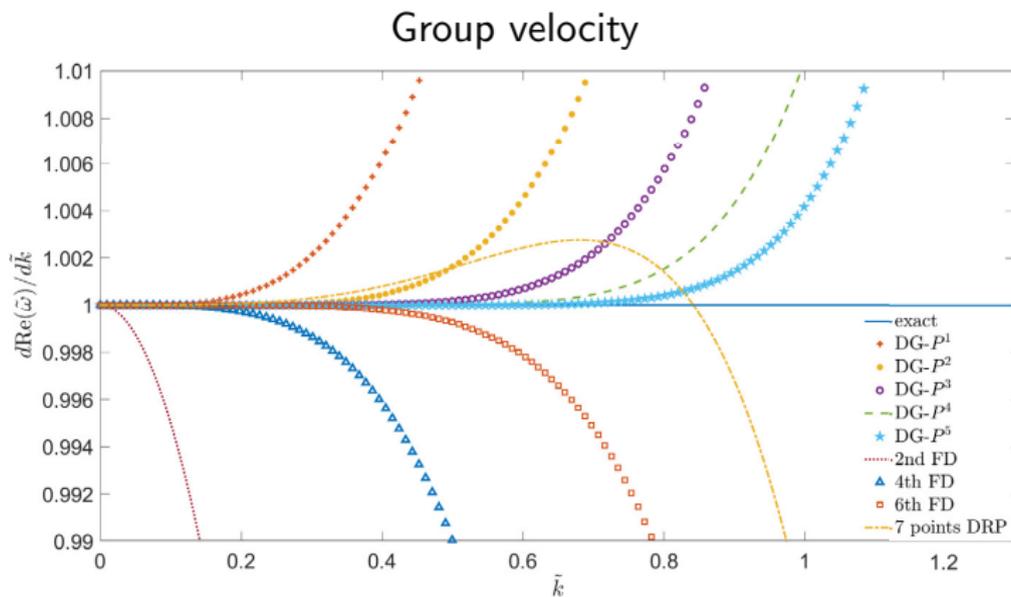


Fig. 3.3. Relative errors of $\text{Re}(\tilde{\omega})$ for DG schemes and central FD schemes.

(Cheng & Shu [14])

Dissipation & dispersion property



(Cheng & Shu [14])

Dissipation & dispersion property

Linear advection: Comprison of FD vs DG

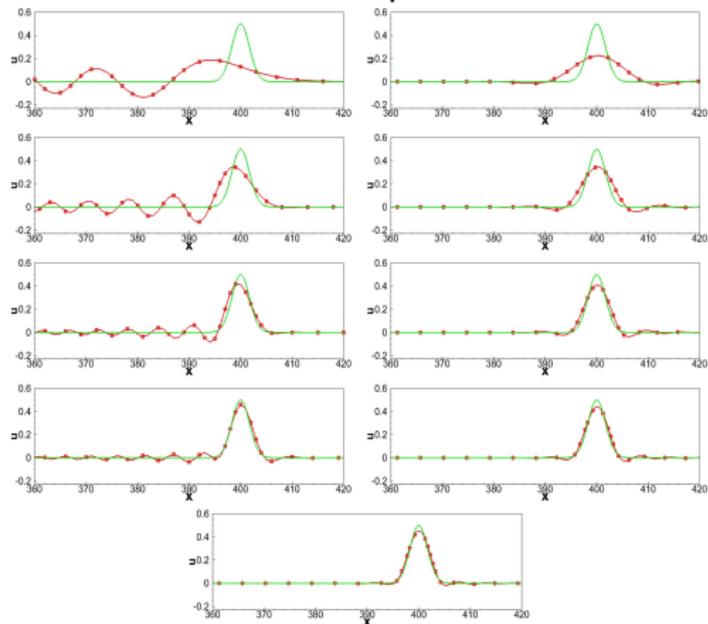


Fig. 6.1. Comparisons between the numerical and exact solutions of the convective wave equation, $b = 2$, $t = 400$, $h_{FD} = 1$, $h_{DG-p} = q + 1$. Lines without square symbols correspond to exact solutions while lines with square symbols correspond to numerical solutions. The subgraphs from top to bottom on the left correspond to FD schemes of order 2, 4, 6, optimized 7-point DRP scheme, and the subgraphs from top to bottom on the right correspond to DG $P^1 - P^4$. The bottom subgraph corresponds to DG- P^5 .

(Cheng & Shu [14])

Dissipation & dispersion property

- Comparison with compact schemes [1]
- One **physical** mode, several **non-physical/parasitic** modes
- Physical mode determines dominant behaviour at low wave-numbers
- All modes contribute to solution behaviour at large wave-numbers
 - ▶ Non-modal analysis [25]
 - ▶ Combined mode analysis [1]

DG in multiple dimensions

Types of basis functions

\mathbb{P}_N : complete polynomials

- Example: $\mathbb{P}_1 = \text{span}\{1, x, y\}$
- Used on triangles/tetrahedra
- Nodal Lagrange (many options) or modal (Jacobi polynomials)

\mathbb{Q}_N : tensor product polynomials

- Example: $\mathbb{Q}_1 = \text{span}\{1, x, y, xy\}$
- Used on quadrilateral/hexahedra
- Nodal Lagrange (GL or GLL) or modal (tensor product of Legendre)

Basis functions defined on mapped elements

Basis functions defined on real elements: Taylor basis [40]

Mappings

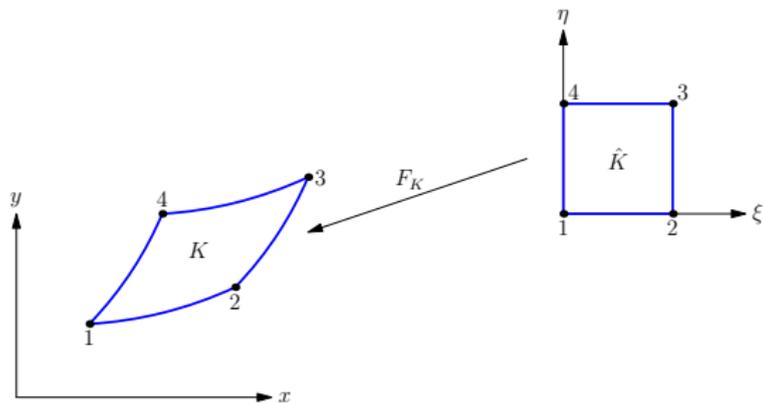
Map element K to reference element \hat{K}

$$\mathbf{x} \in K, \quad \boldsymbol{\xi} \in \hat{K}$$

$$F_K : \hat{K} \rightarrow K$$

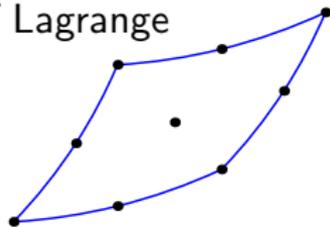
$$\mathbf{x} = F_K(\boldsymbol{\xi})$$

E.g., $\hat{K} = [0, 1] \times [0, 1]$ if K is a quadrilateral



- K triangle/tetrahedron: F_K is an affine map
- K quadrilateral/hexahedra: F_K is bi/tri-linear map
- K can be curved element: F_K given in terms of Lagrange polynomials of degree $M \geq 1$

$$\mathbf{x} = \sum_{i=0}^M \sum_{j=0}^M \mathbf{x}_{ij} l_i(\xi) l_j(\eta)$$



DG for 2-D NS

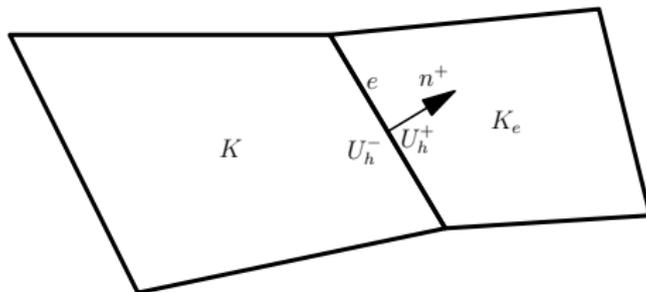
$$\partial_t \mathbf{U} + \nabla \cdot \mathbf{F}(\mathbf{U}) = \nabla \cdot \mathbf{G}(\mathbf{U}, \nabla \mathbf{U})$$

Write as first order system

$$\partial_t \mathbf{U} + \nabla \cdot \mathbf{F}(\mathbf{U}) = \nabla \cdot \mathbf{G}(\mathbf{U}, \mathbf{Q}), \quad \mathbf{Q} = \nabla \mathbf{U}$$

Inside each element K , approximate by degree N polynomials

$$\mathbf{U}_h = \sum_{i=0}^N \sum_{j=0}^N \mathbf{U}_{ij}^K \phi_i(\xi) \phi_j(\eta), \quad \mathbf{Q}_h = \sum_{i=0}^N \sum_{j=0}^N \mathbf{Q}_{ij}^K \phi_i(\xi) \phi_j(\eta)$$



DG for 2-D NS

Multiply by test function Φ_h and integrate on element K

$$\begin{aligned} & \int_K (\Phi_h \partial_t \mathbf{U}_h - \mathbf{F}(\mathbf{U}_h) \cdot \nabla \Phi_h + \mathbf{G}(\mathbf{U}_h, \mathbf{Q}_h) \cdot \nabla \Phi_h) d\mathbf{x} \\ & + \sum_{e \in \partial K_i} \int_e \Phi_h^- [\mathbf{F}(\mathbf{U}_h^-, \mathbf{U}_h^+, \mathbf{n}^+) - \mathbf{G}(\mathbf{U}_h^-, \mathbf{Q}_h^-, \mathbf{U}_h^+, \mathbf{Q}_h^+, \mathbf{n}^+)] ds \\ & + \sum_{e \in \partial K_b} \int_e \Phi_h [\mathbf{F}(\mathbf{U}_h, \mathbf{U}_b, \mathbf{n}) - \mathbf{G}(\mathbf{U}_h, \mathbf{Q}_h, \mathbf{U}_b, \mathbf{Q}_h, \mathbf{n})] ds = 0 \end{aligned}$$

$$\begin{aligned} & \int_K (\Phi_h \mathbf{Q}_h + \mathbf{U}_h \nabla \Phi_h) d\mathbf{x} - \sum_{e \in \partial K_i} \int_e \mathbf{U}(\mathbf{U}_h^-, \mathbf{U}_h^+) \Phi_h^- \mathbf{n}^+ ds \\ & - \sum_{e \in \partial K_b} \int_e \mathbf{U}_b \Phi_h \mathbf{n} ds = 0 \end{aligned}$$

Each Φ_h is of the form $\phi_i(\xi)\phi_j(\eta)$, $0 \leq i, j \leq N$

$F(U_h^-, U_h^+, \mathbf{n}^+)$: inviscid numerical flux

BR1 scheme

$$U(U_h^-, U_h^+) = \frac{1}{2}(U_h^- + U_h^+)$$

$$G(U_h^-, Q_h^-, U_h^+, Q_h^+, \mathbf{n}^+) = \frac{1}{2}[G(U_h^-, Q_h^-, \mathbf{n}^+) + G(U_h^+, Q_h^+, \mathbf{n}^+)]$$

Gassner et al. [30] show stability of this scheme for GLL nodes.

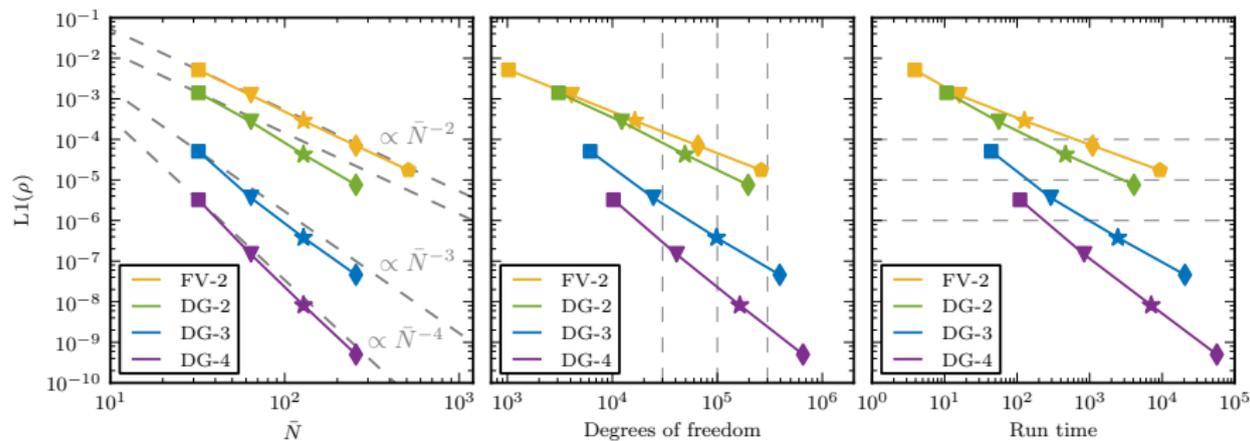
LDG scheme

$$U(U_h^-, U_h^+) = U_h^-$$

$$G(U_h^-, Q_h^-, U_h^+, Q_h^+, \mathbf{n}^+) = G(U_h^+, Q_h^+, \mathbf{n}^+)$$

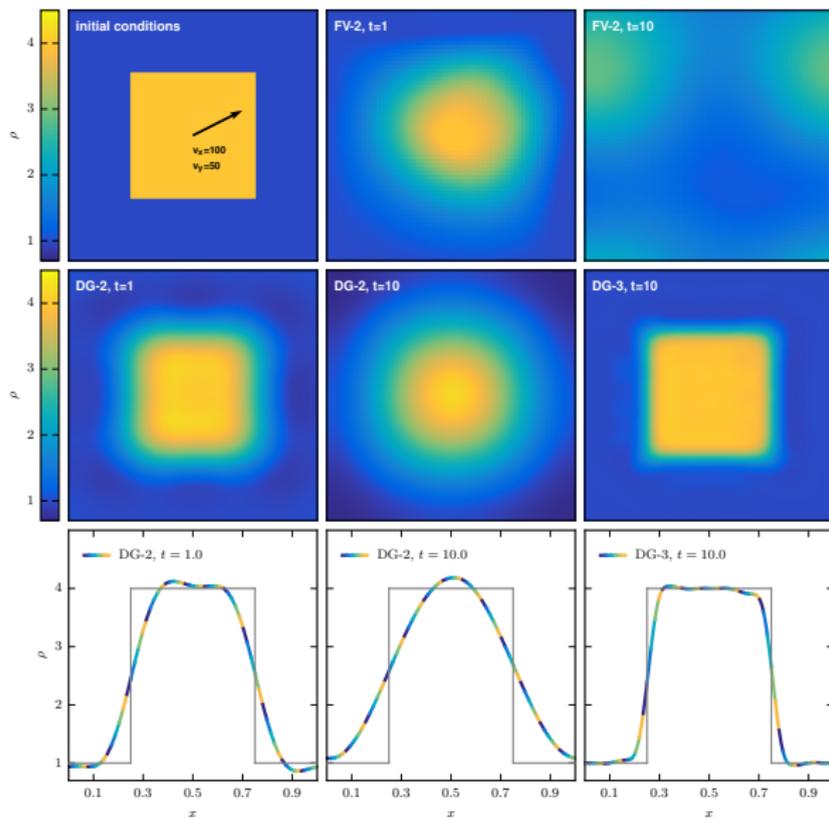
Time stepping: Explicit RK or implicit schemes

Inviscid isentropic vortex

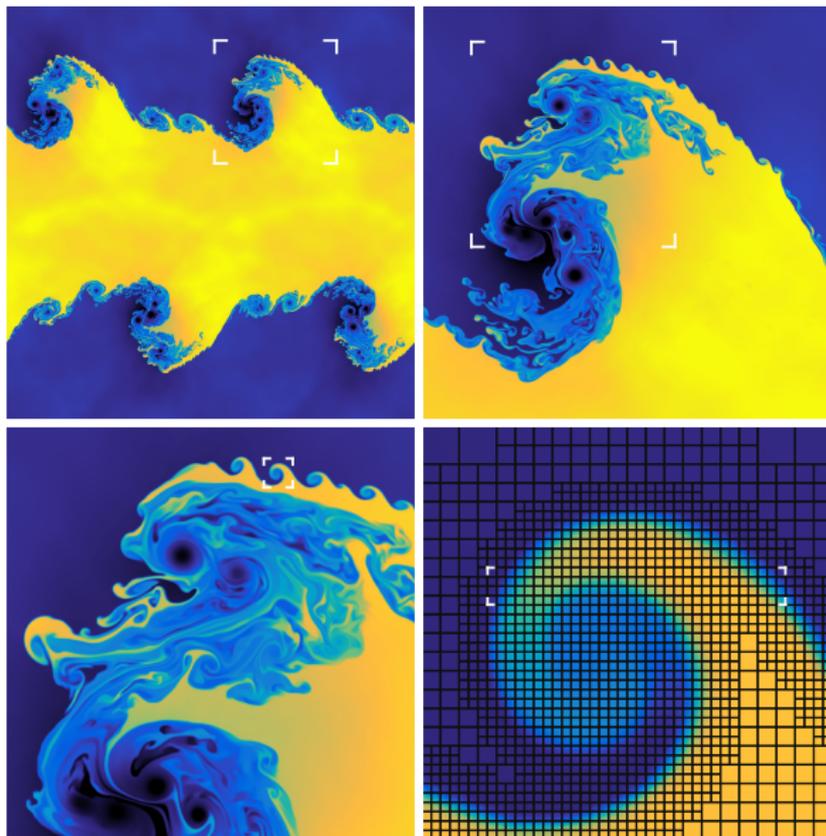


(Schaal et al. [44])

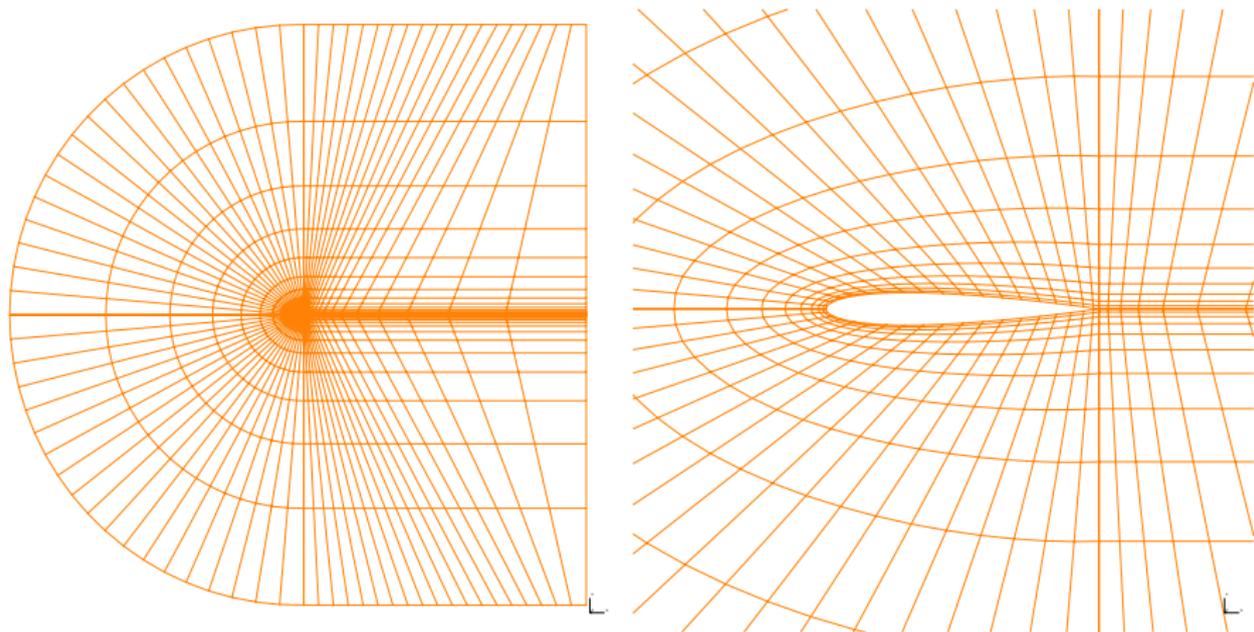
Advection of square profile (Schaal et al. [44])



Inviscid Kelvin-Helmholtz with AMR (Schaal et al. [44])



NACA0012 airfoil



Boundary approximation

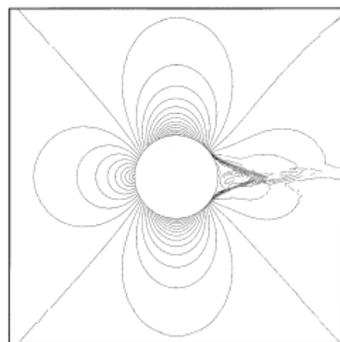
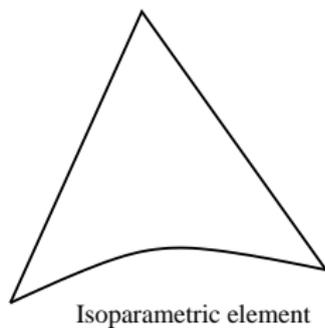
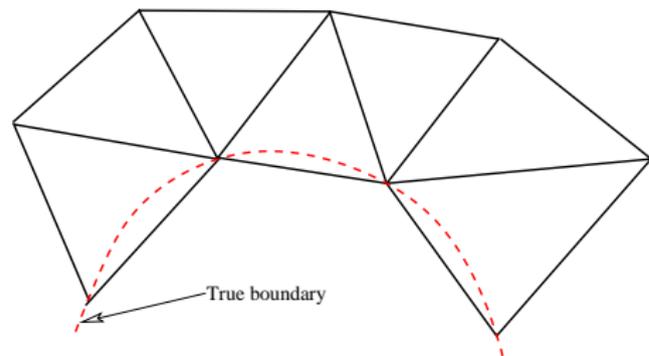


FIG. 8. Mach isolines around a circle with P1Q1 elements on the 128×32 grid.

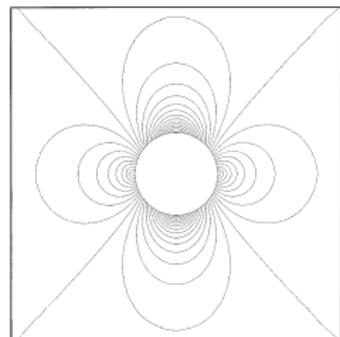
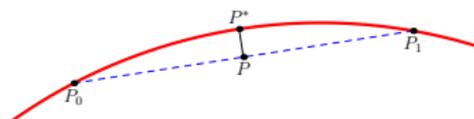
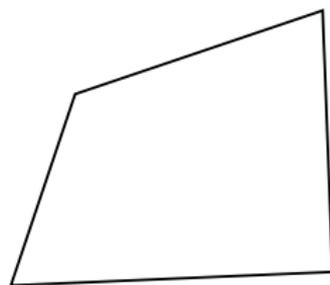


FIG. 12. Mach isolines around a circle with P1Q2 elements on the 128×32 grid.

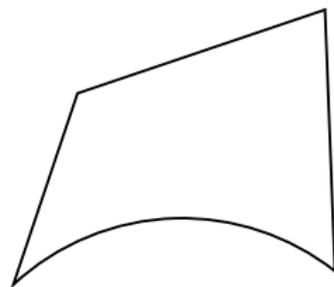
(Bassi & Rebay [6])

High order meshes

Start with Q1 mesh



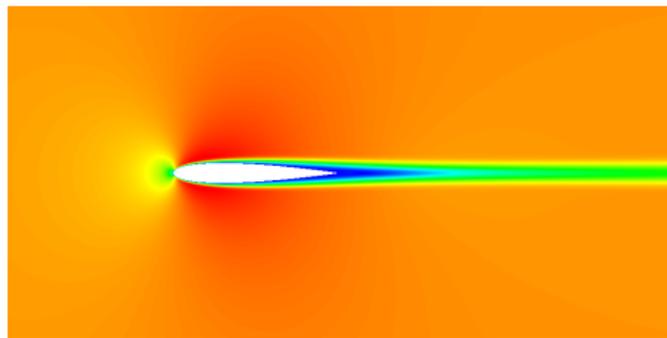
Normal projection of straight edge onto curved boundary



Only boundary edges are high order

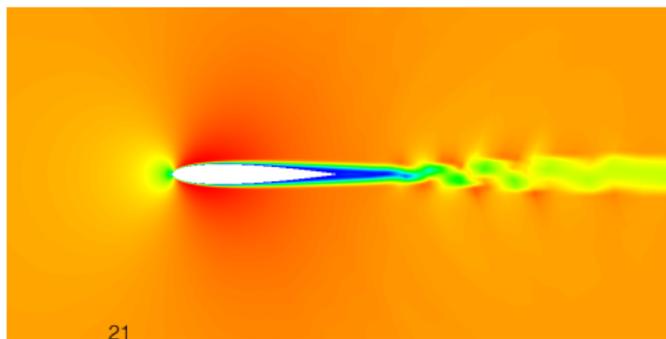
NACA0012 airfoil, $M=0.5$, $AOA=0$ deg.

$N=3$, 1444 cells



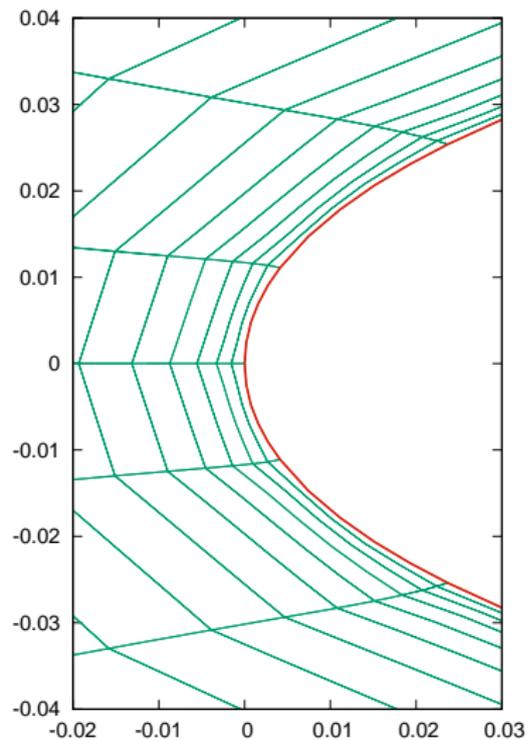
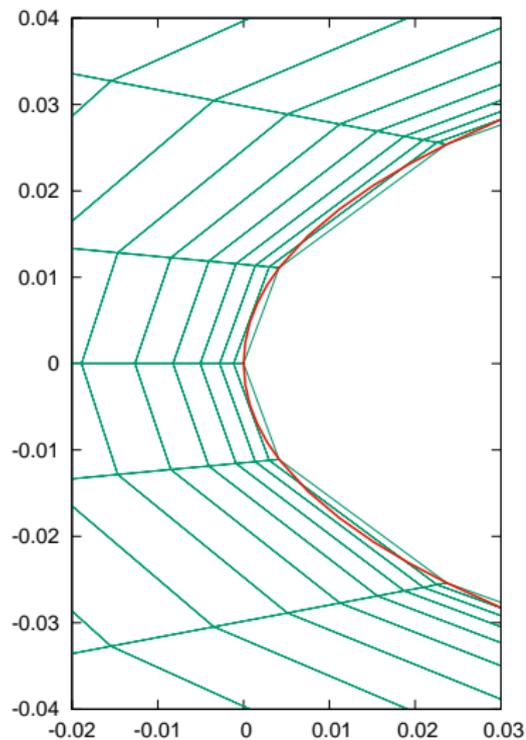
$Re=5000$

$Re=10000$



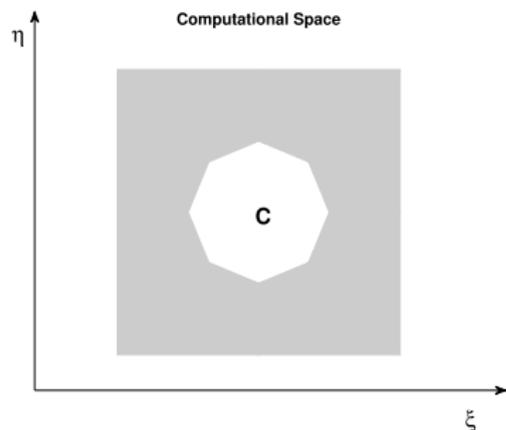
Wake instability due to compressibility effects [8]

High order meshes

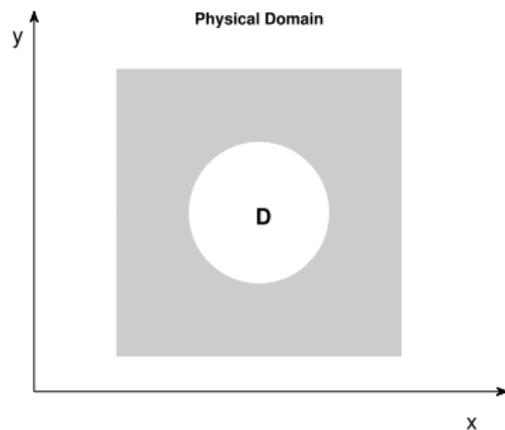


Linear mesh smoothed using Winslow equations

Winslow-based mapping²



$$\longrightarrow$$
$$x=x(\xi,\eta)$$
$$y=y(\xi,\eta)$$



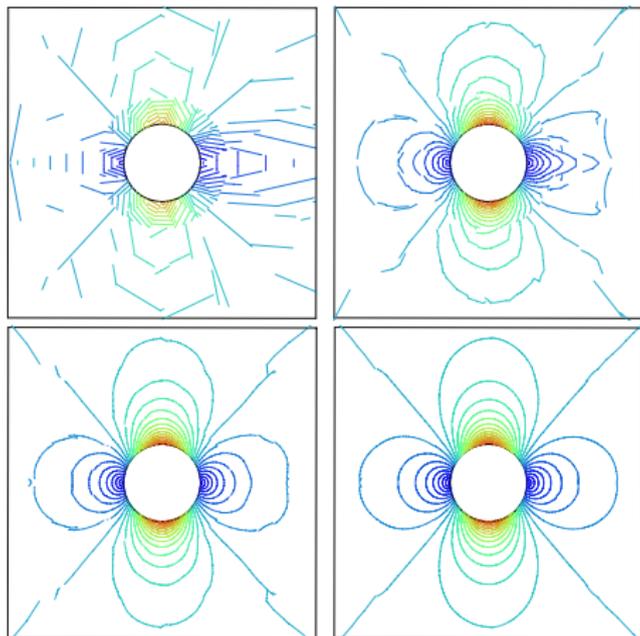
$$g^{ij} \partial_i \partial_j x_k = 0 \quad \text{for } k = 1, \dots, n,$$

$$g^{ij} g_{jk} = \delta_{ik} \quad \text{and} \quad g_{ij} = \partial_i x_k \partial_j x_k$$

$$\begin{aligned} \partial_i (g^{ij}) + \alpha_j &= 0, & \text{for } j = 1, \dots, n, \\ \partial_i (g^{ij} \partial_j x_k) + \alpha_j \partial_j x_k &= 0, & \text{for } k = 1, \dots, n. \end{aligned}$$

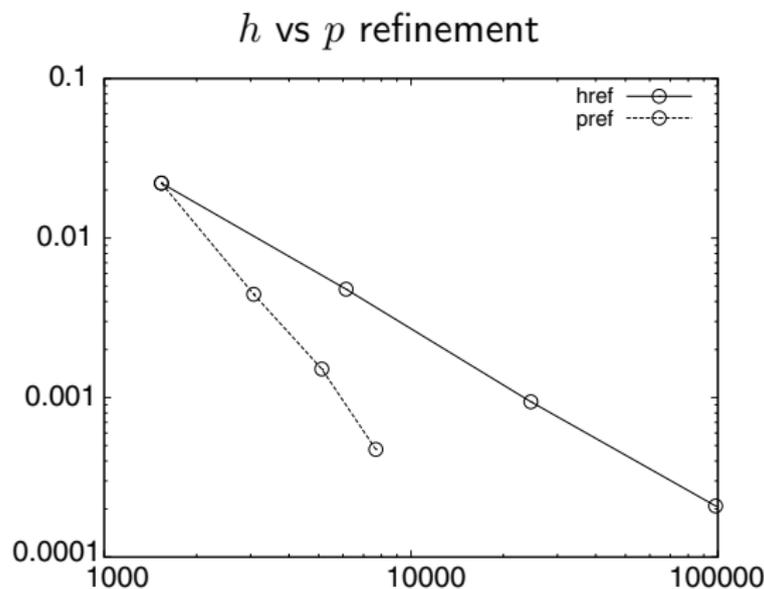
²Persson and Fortunato [27]

Inviscid flow over cylinder



16×4 mesh, degrees 1, 2, 3, 4 (Krivodonova & Berger [37])

Inviscid flow over cylinder



L_2 error of total pressure on cylinder surface as function of number of degrees of freedom (Krivodonova & Berger [37])

Inviscid flow over ellipse

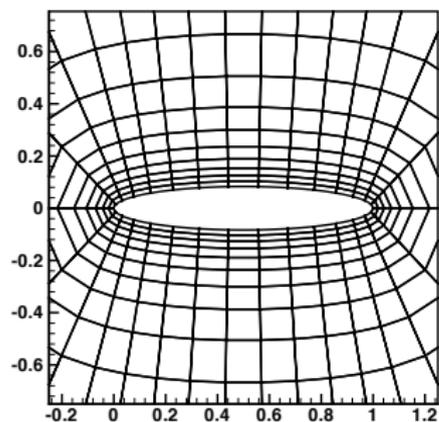


Figure 8. Closeup of Mesh 1 for the ellipse problem.

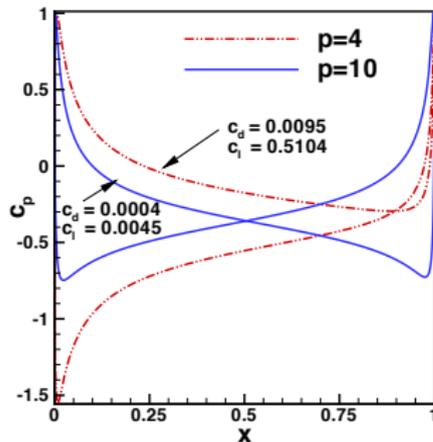
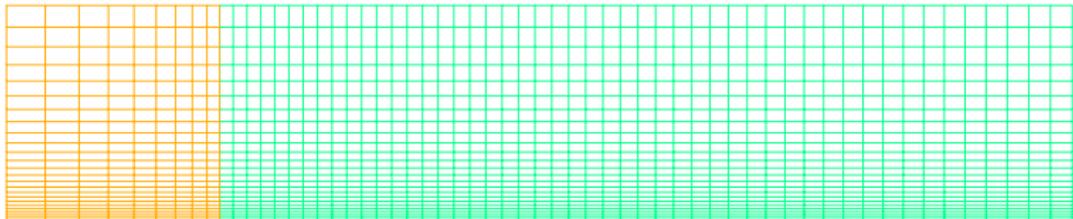


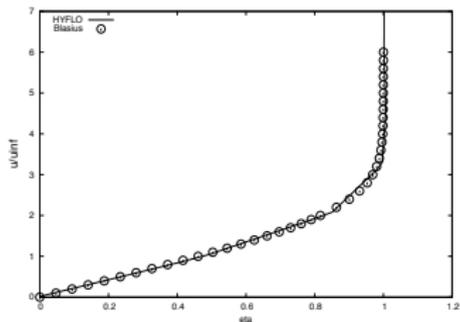
Figure 9. Pressure coefficient distribution for both Mesh 1 ($p = 4$) and Mesh 2 ($p = 10$).

(Collis & Ghayour, 2003)

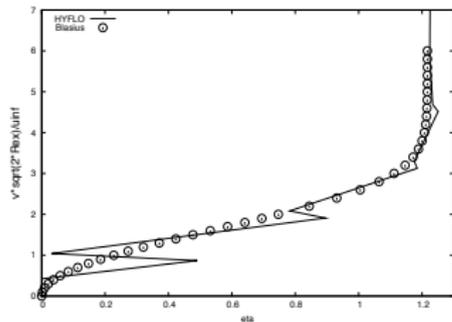
Laminar boundary layer



1392 cells, $Re=1.5e5$, $Mach=0.1$

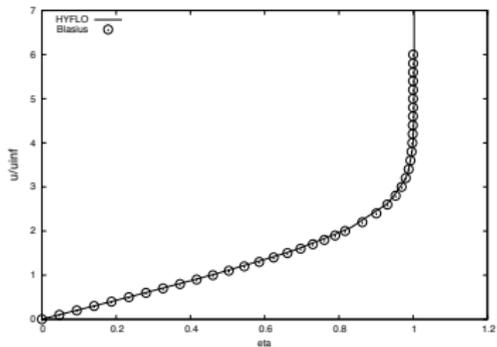


$N=1$

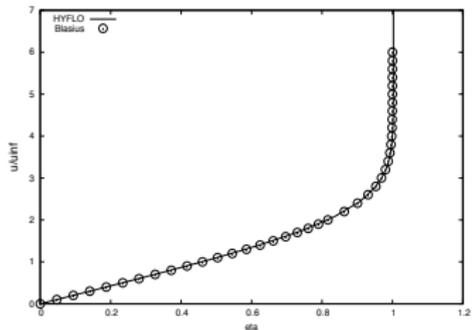
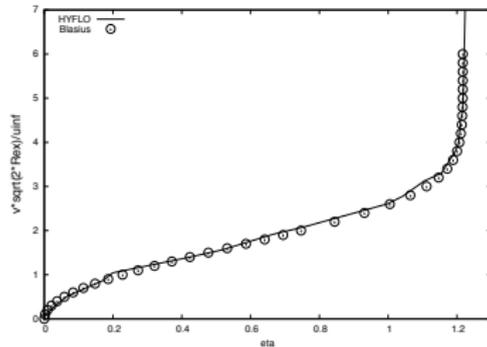


HYFLO

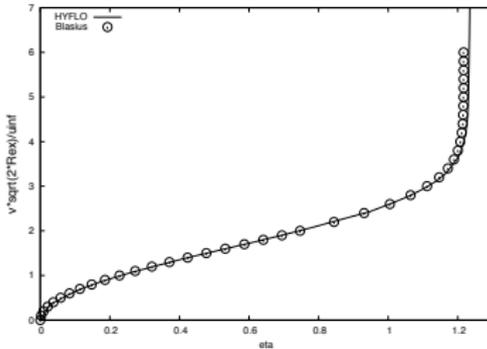
Laminar boundary layer



$N=2$



$N=3$



HYFLO

Boundary conditions

- Imposed weakly through the fluxes
- No-slip bc also imposed weakly
Gives better accuracy and stability [20]
- Farfield bc can be challenging
 - ▶ wake going upto the boundary
 - ▶ Non-reflecting bc much harder to device
 - ▶ Sponge layers can be used [28], [7], [21], [26]
- Example movie: flow past cylinder

Entropy variable based schemes

Convex entropy function: $S(\mathbf{U}) = -\frac{\rho s}{\gamma-1}$, $s = \ln(p/\rho^\gamma)$

Entropy variables: $\mathbf{V} = \frac{\partial S}{\partial \mathbf{U}}$

Used long ago by Deshpande et al. (q-KFVS [32], q-LSKUM [22])

NS in entropy variables

$$\underbrace{\frac{\partial \mathbf{U}}{\partial \mathbf{V}}}_{spd} \frac{\partial \mathbf{V}}{\partial t} + \underbrace{\frac{\partial \mathbf{F}_\alpha}{\partial \mathbf{V}}}_{sym} \frac{\partial \mathbf{V}}{\partial x_\alpha} = \frac{\partial}{\partial x_\alpha} \left(D_{\alpha\beta} \frac{\partial \mathbf{V}}{\partial x_\beta} \right)$$

$$D = [D_{\alpha\beta}] = D^\top \geq 0$$

Take dot product with \mathbf{V} to get entropy inequality

$$\frac{\partial S}{\partial t} + \nabla \cdot (\mathbf{v}S) = \frac{\partial}{\partial x_\alpha} \left(\mathbf{V}^\top D_{\alpha\beta} \frac{\partial \mathbf{V}}{\partial x_\beta} \right) - \underbrace{\left(\frac{\partial \mathbf{V}}{\partial x_\alpha} \right)^\top D_{\alpha\beta} \frac{\partial \mathbf{V}}{\partial x_\beta}}_{\geq 0}$$

This property can be mimicked in a DG scheme.

Entropy variable based schemes

Two ingredients are necessary.

- Need entropy conservative fluxes
 - ▶ Euler equations: Ismail/Roe [36], Chandrashekar [11]
 - ▶ Ideal MHD: Chandrashekar/Klingenberg [12], Winters et al. [46]
- Need exact quadrature to do integration-by-parts
 - ▶ Summation-by-parts property [31], [30]

⇒ Semi-discrete entropy stability for any order of accuracy !!!

Beneficial for under-resolved LES and DNS computations [29], [47]

Kinetic energy and/or entropy conserving schemes

⇒ behave like central schemes

⇒ add explicit SGS model or filtering

For some details, see my lecture [slides](#)

Turbulent flows: DG + RANS

- $k - \omega$ model: [4], [39]
- SA model: [24], [39]
- Very high order may not give much improvement
 \implies limitations of RANS model may play bigger role
- Goal-based grid adaptation
 - ▶ Finite element facilitates adjoint approach

Turbulent flows: μ DNS/ILES

- Inherent dissipation in Riemann solvers acts as implicit SGS model
- Notable successes in computing turbulent and transitional flows
- Instability at very high orders [47]
 - \implies inherent dissipation may not be enough
 - \implies integration/aliasing errors
- Kinetic energy/entropy preserving schemes, with SBP property may help

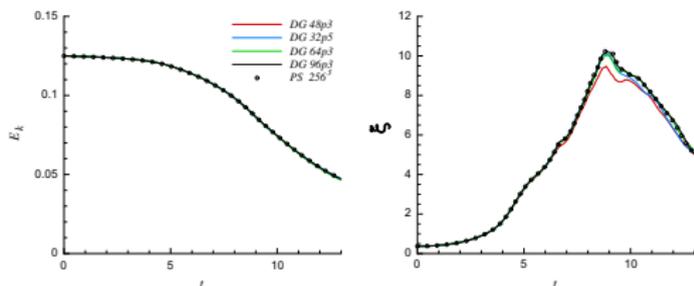


Fig. 10. Temporal evolution of mean energy and entropy for the TGV computations at $Re = 1600$.

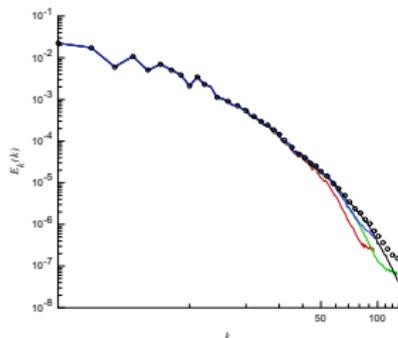


Fig. 11. Energy spectra at $t = 8.2$ for the TGV computations at $Re = 1600$. Same legend as Fig. 11.

(Chapelier et al. [13])

DG for ideal MHD (Guillet et al. [33])

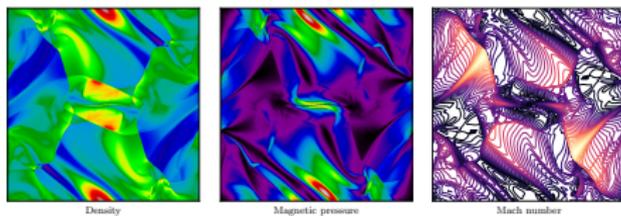


Figure 10. Orszag-Tang vortex test problem at $t = 0.5$. The density, pressure and Mach number are shown on a 512^2 grid, computed using the third-order DG scheme with the Powell method.

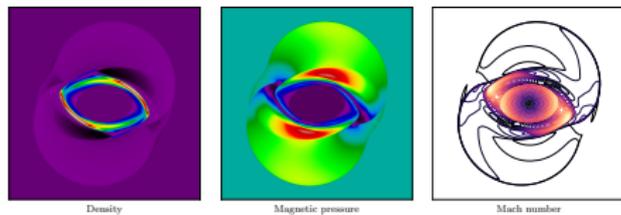


Figure 16. Magnetic rotor test problem. The density, pressure and Mach number contours in the 2D magnetic adiabatic rotor test are shown, on a 512^2 grid using the third-order Powell scheme.

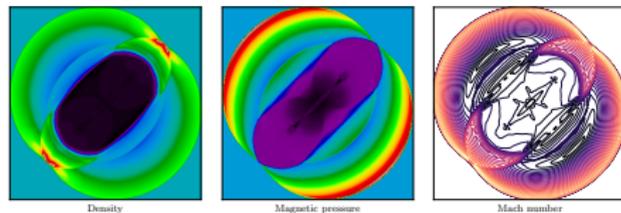
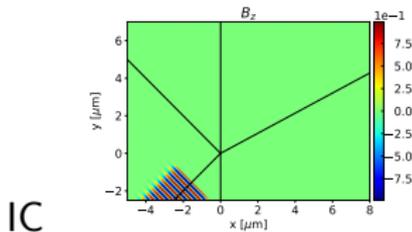
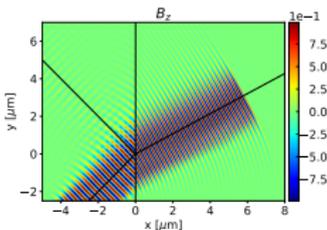


Figure 20. Two-dimensional MHD blast test problem. The density, magnetic pressure and Mach number contours are shown on a 256^2 grid using the third-order Powell scheme.

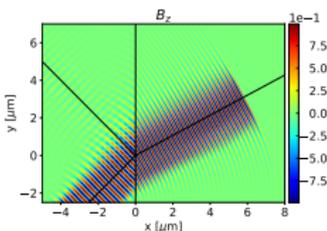
DG for Maxwell equations (Hazra et al. [35])



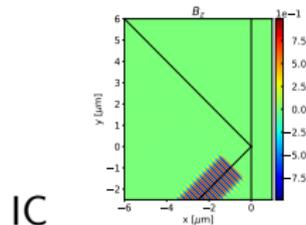
$k = 3$



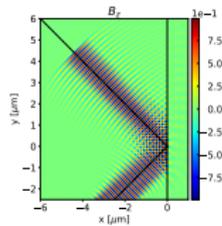
$k = 4$



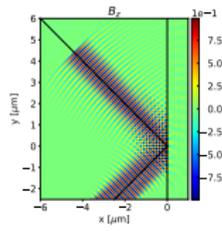
Refraction of compact electromagnetic beam by dielectric slab, 650×475 cells



$k = 3$



$k = 4$



Total internal reflection of compact electromagnetic beam by dielectric slab, 350×425 cells

Summary: Advantages

- Very high orders, spectral accuracy
- Low dissipation/dispersion errors
- Good candidate for computing
 - ▶ multi-scale phenomena, turbulent flows
 - ▶ vortex dominated flows
 - ▶ aero-acoustics
- Error independent of time upto $t = O(1/h)$
useful for long time simulations
- DG is ideal for unstructured grids, hybrid elements
- Local grid refinement: h and p refinement
- Grids with hanging nodes, quadtree/octree grids

Summary: Issues

- High order meshes essential
- Effect of quadrature on stability
- Effect of inviscid numerical flux [42]
 - ⇒ both under/over diffusion is harmful
 - ⇒ Roe-type schemes to be preferred
 - ⇒ upwind schemes may add too much diffusion at low mach
- Good artificial boundary conditions
- Efficient implementation
 - matrix-free, sum factorizations [38]
- Transonic/supersonic turbulence
 - ▶ shock dominated
 - ▶ need limiters or artificial diffusion: effect on accuracy ?

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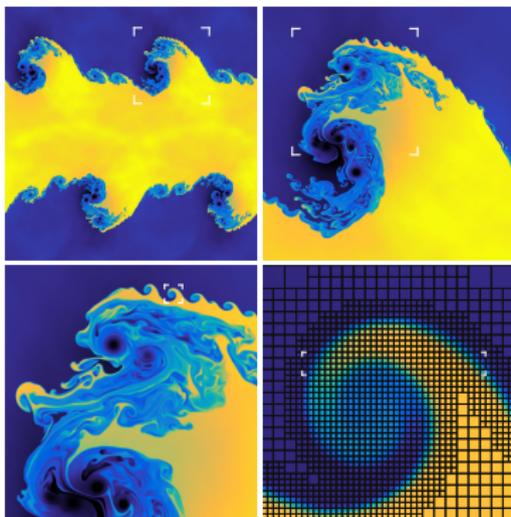
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Thank You